

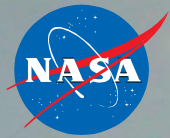
Estimating Flow-Through Balance Momentum Tares with CFD

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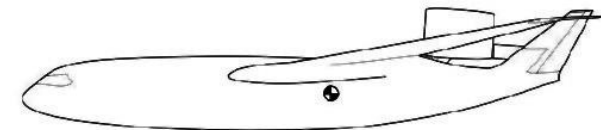
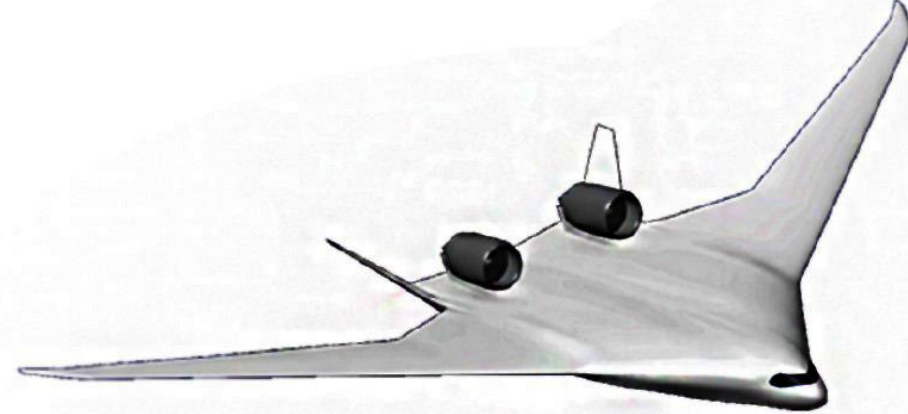
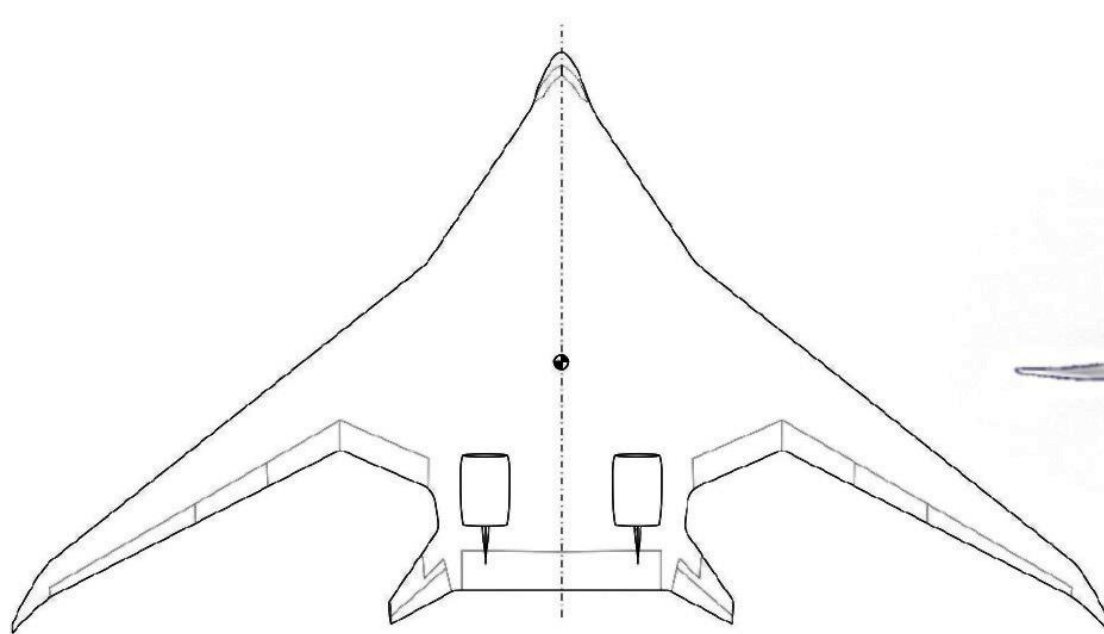
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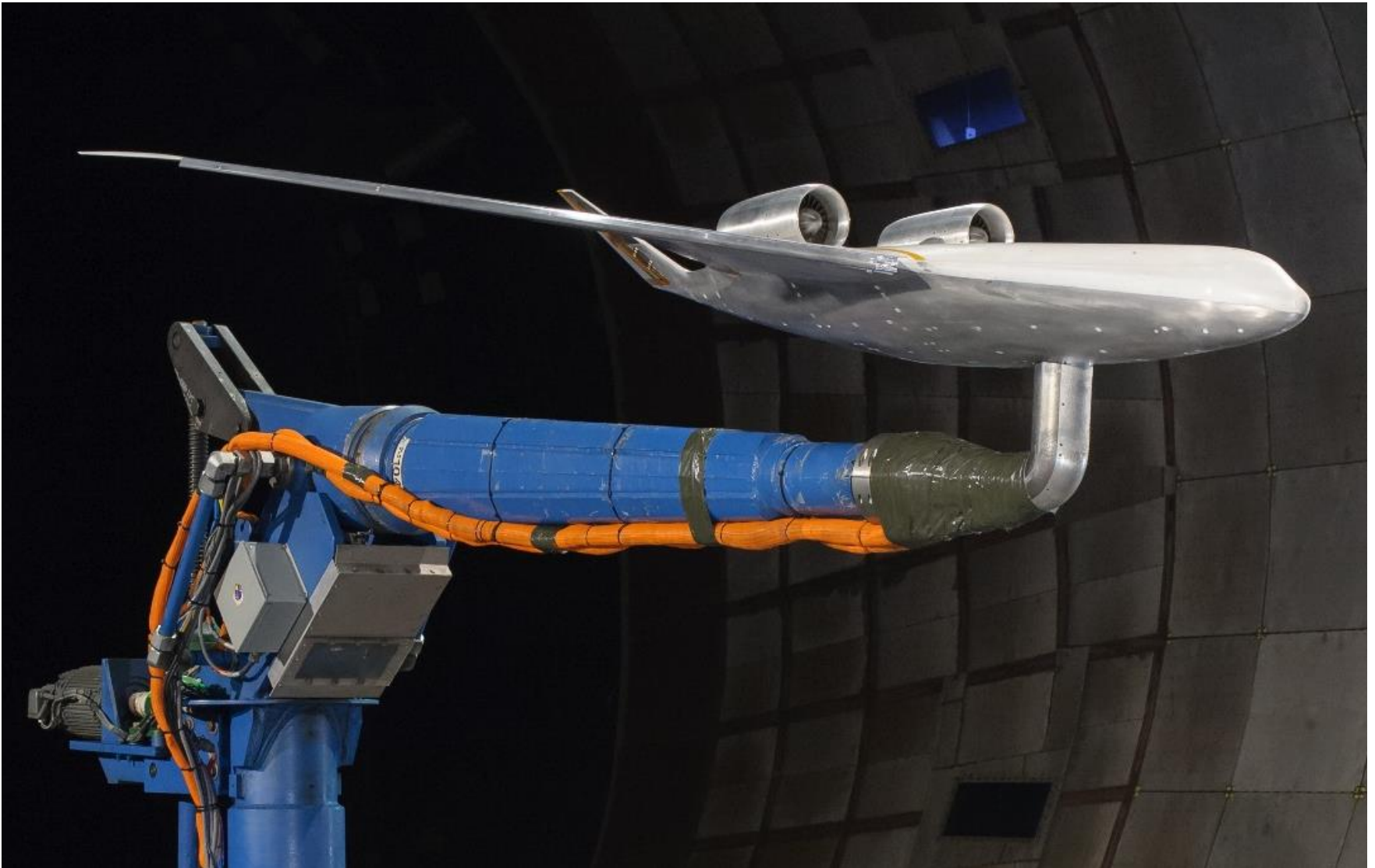


- HWB and Flow-through balance (FTB)
- Control volume approach for FTB
- Momentum tare calculations
- CFD simulations with STAR-CCM+
- Comparisons with FTB calibrations
- Simulations for HWB with TPS in 40x80
- Recommendations for future FTB testing

HWB Configuration

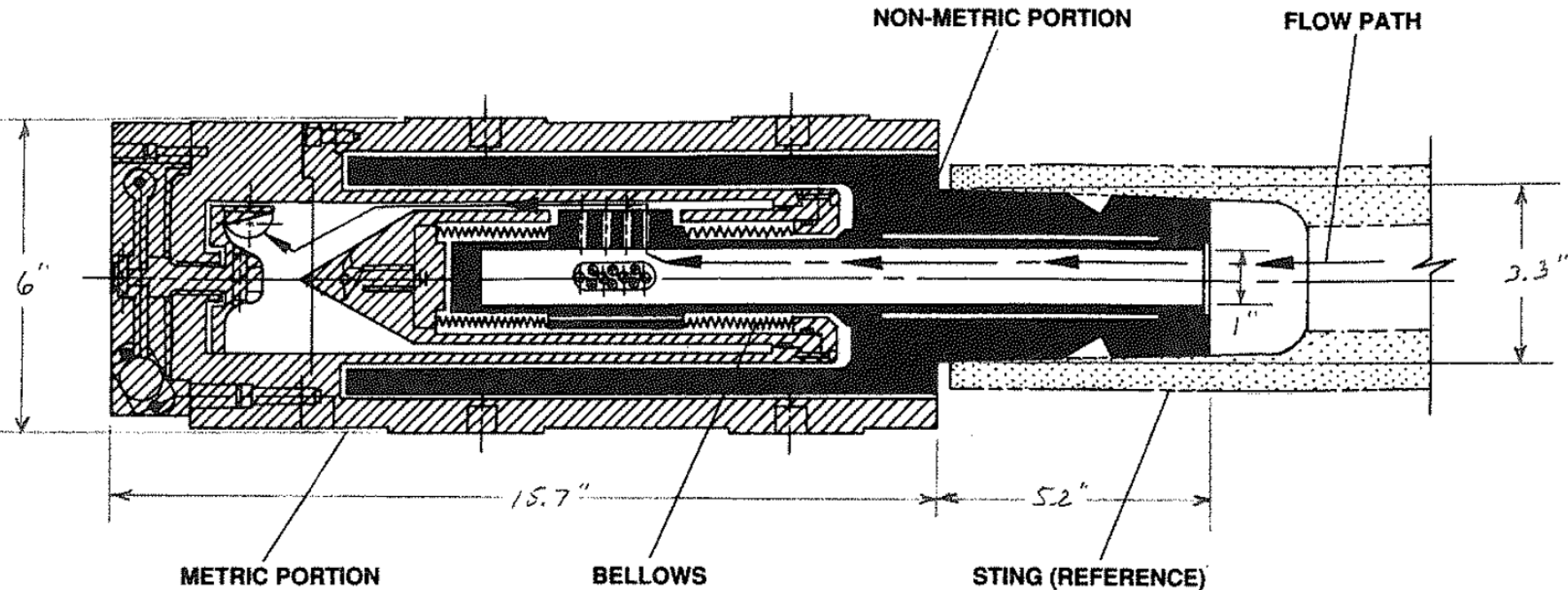


HWB 40x80 Installation with TPS units

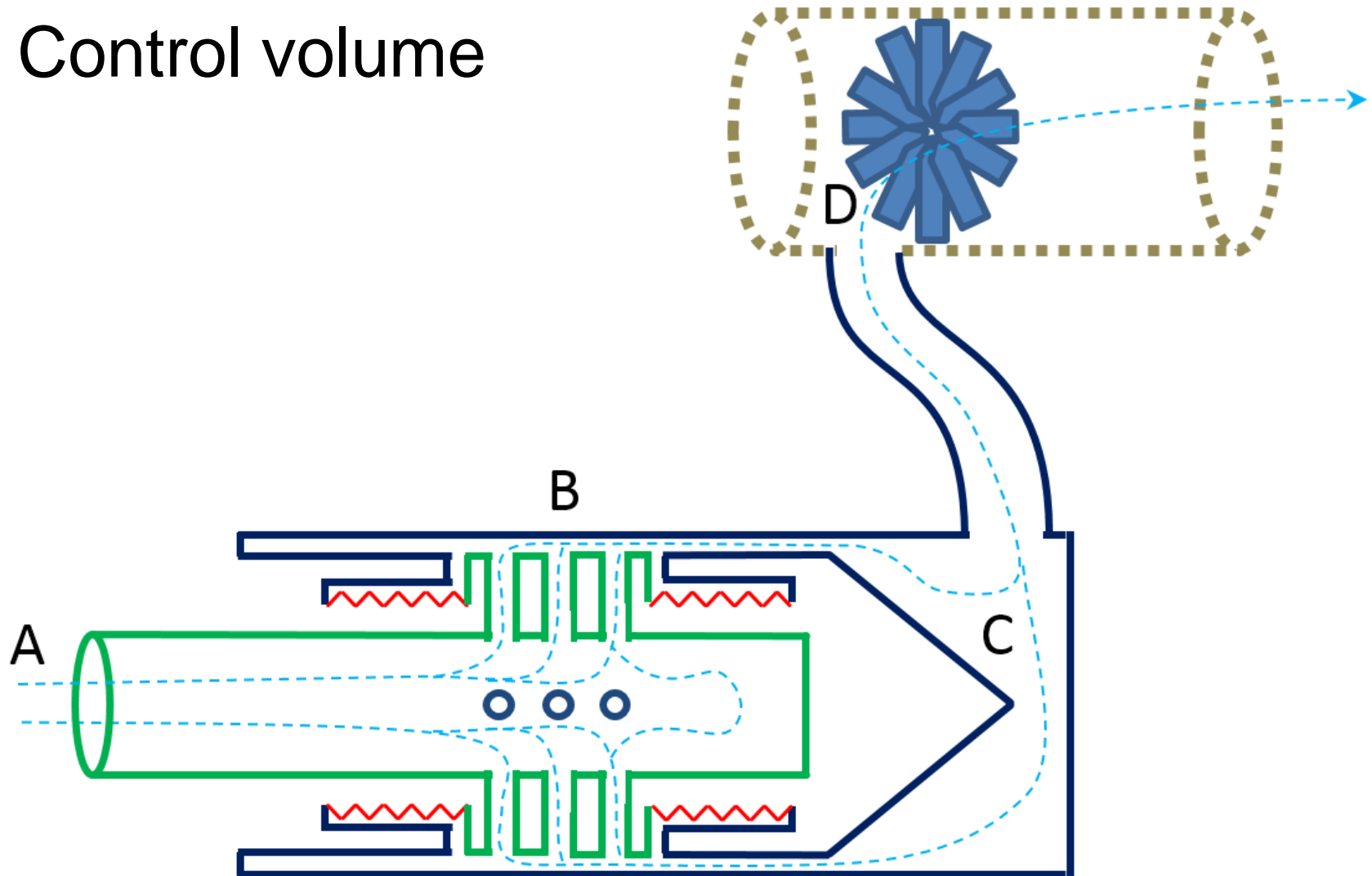


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- The diagram illustrates the TPS Unit and Model Support Sting. The TPS Unit is shown as a blue star-like component within a dashed yellow rectangle. It is connected by a blue line to a blue rectangular component on the Model Support Sting. The Model Support Sting is a green line that runs along the Vehicle External Surface, which is indicated by a dashed purple line. The Model Support Sting is connected to the TPS Unit and the blue rectangular component.

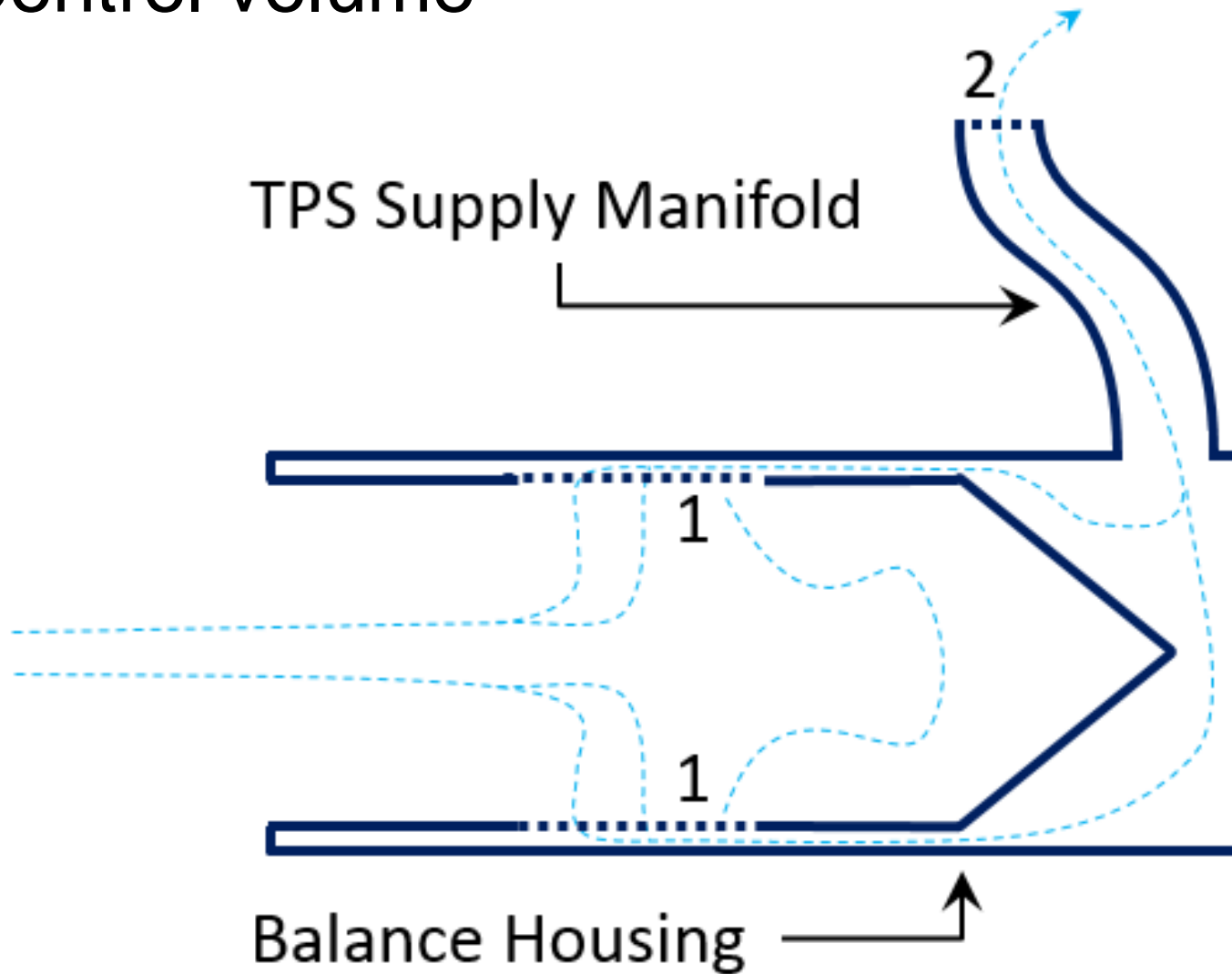
- Flow-through balance internal details
 - Designed and patented by ARC, 1988



- Control volume



- Control volume



- Momentum tare calculation approach
 - Assume steady flow, inertial frame
 - Momentum change computed by flux across inlets and exits
 - Equivalent to restraining forces on CV

$$\Delta \mathbf{Mom} = \iint_{CV} \mathbf{V} (\mathbf{V} \cdot d\mathbf{A}) = \sum \mathbf{F}_{CV} = \mathbf{F}_{bellows} + \mathbf{F}_{pressure} + \mathbf{F}_{shear} = \mathbf{F}_{bellows} - \iint_{CV} (p - p_{\infty}) d\mathbf{A} + \iint_{CV} d\mathbf{F}_{shear}$$

$$\iint_{inlet} \mathbf{V} (\mathbf{V} \cdot d\mathbf{A}) + \iint_{exit} \mathbf{V} (\mathbf{V} \cdot d\mathbf{A}) = - \iint_{walls} (p - p_{\infty}) d\mathbf{A} - \iint_{inlet} (p - p_{\infty}) d\mathbf{A} - \iint_{exit} (p - p_{\infty}) d\mathbf{A} + \iint_{walls} d\mathbf{F}_{shear}$$

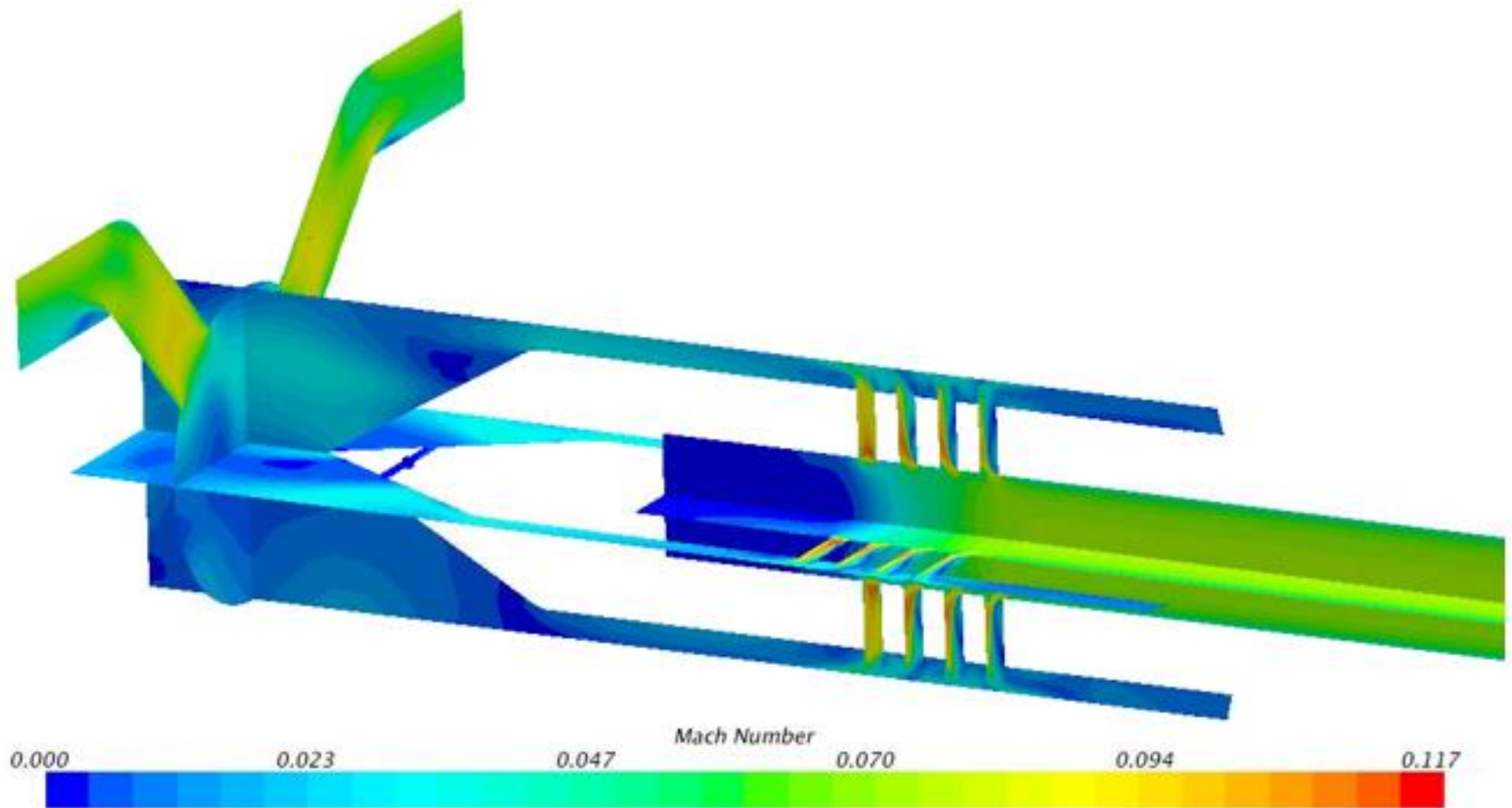
$$\mathbf{T}_{inlet} + \mathbf{T}_{exit} = \iint_{inlet} \mathbf{V} (\mathbf{V} \cdot d\mathbf{A}) + \iint_{inlet} (p - p_{\infty}) d\mathbf{A} + \iint_{exit} \mathbf{V} (\mathbf{V} \cdot d\mathbf{A}) + \iint_{exit} (p - p_{\infty}) d\mathbf{A} = - \iint_{walls} (p - p_{\infty}) d\mathbf{A} + \iint_{walls} d\mathbf{F}_{shear}$$

- Momentum tare calculation approach
 - Tare can be calculated using integration over geometrically simple inlet and exit faces
 - Thrust formulation provides a check on wall pressure and shear integrations

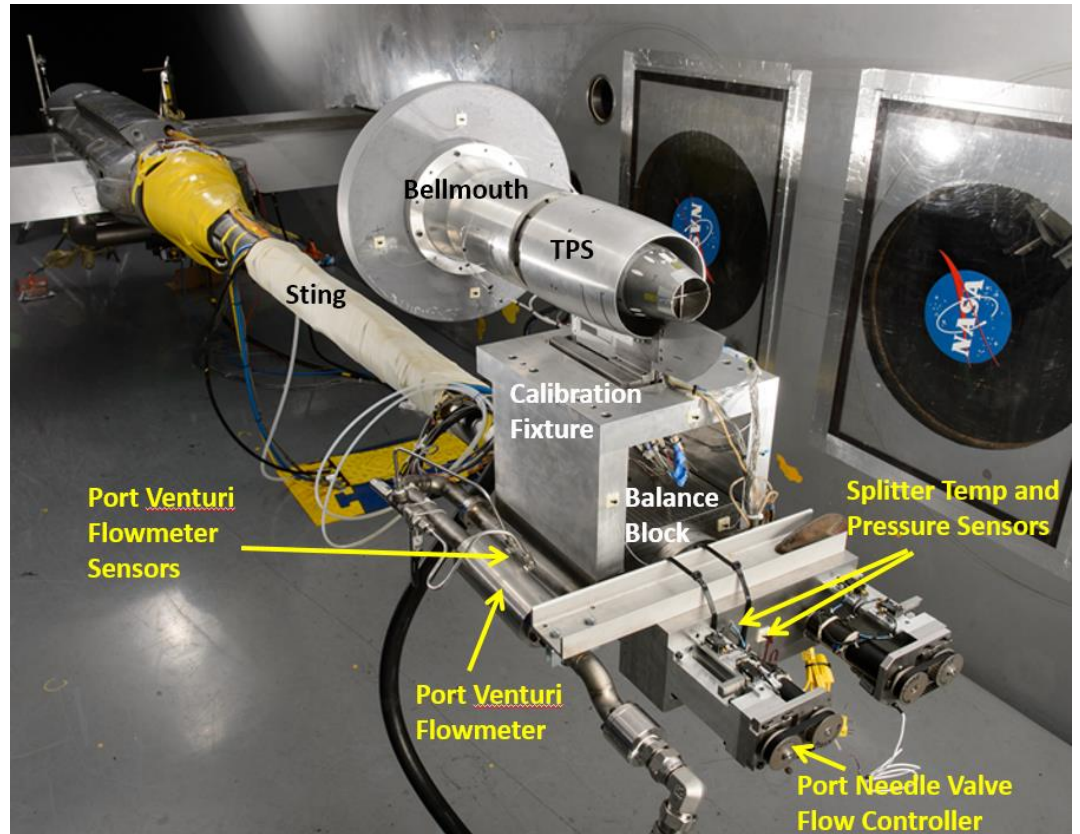
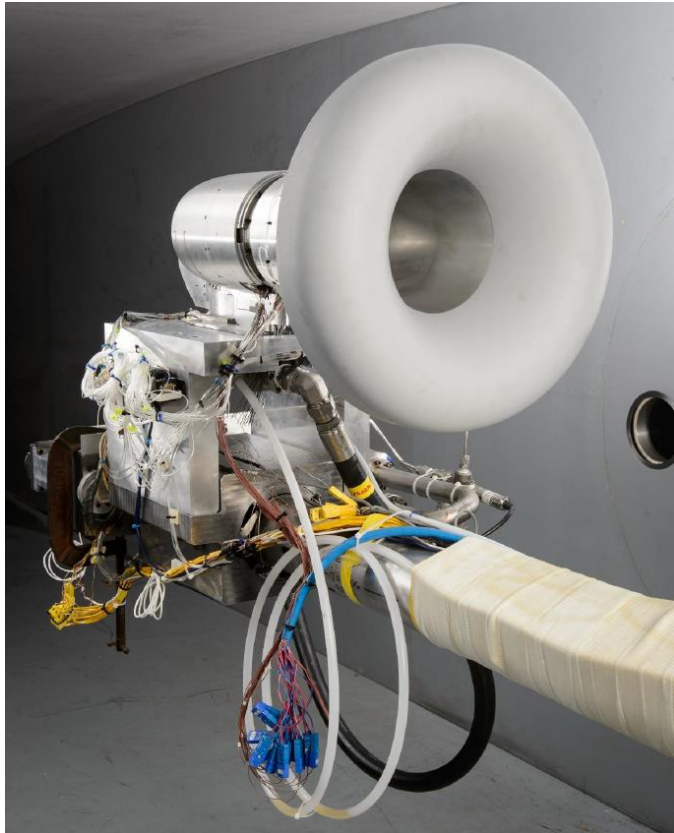
$$\mathbf{T}_{inlet} + \mathbf{T}_{exit} = \iint_{inlet} \mathbf{V} (\mathbf{V} \cdot d\mathbf{A}) + \iint_{inlet} (p - p_{\infty}) d\mathbf{A} + \iint_{exit} \mathbf{V} (\mathbf{V} \cdot d\mathbf{A}) + \iint_{exit} (p - p_{\infty}) d\mathbf{A} = - \iint_{walls} (p - p_{\infty}) d\mathbf{A} + \iint_{walls} d\mathbf{F}_{shear}$$

- Momentum tare calculation approach
- Good match between mass flow and supply pressure

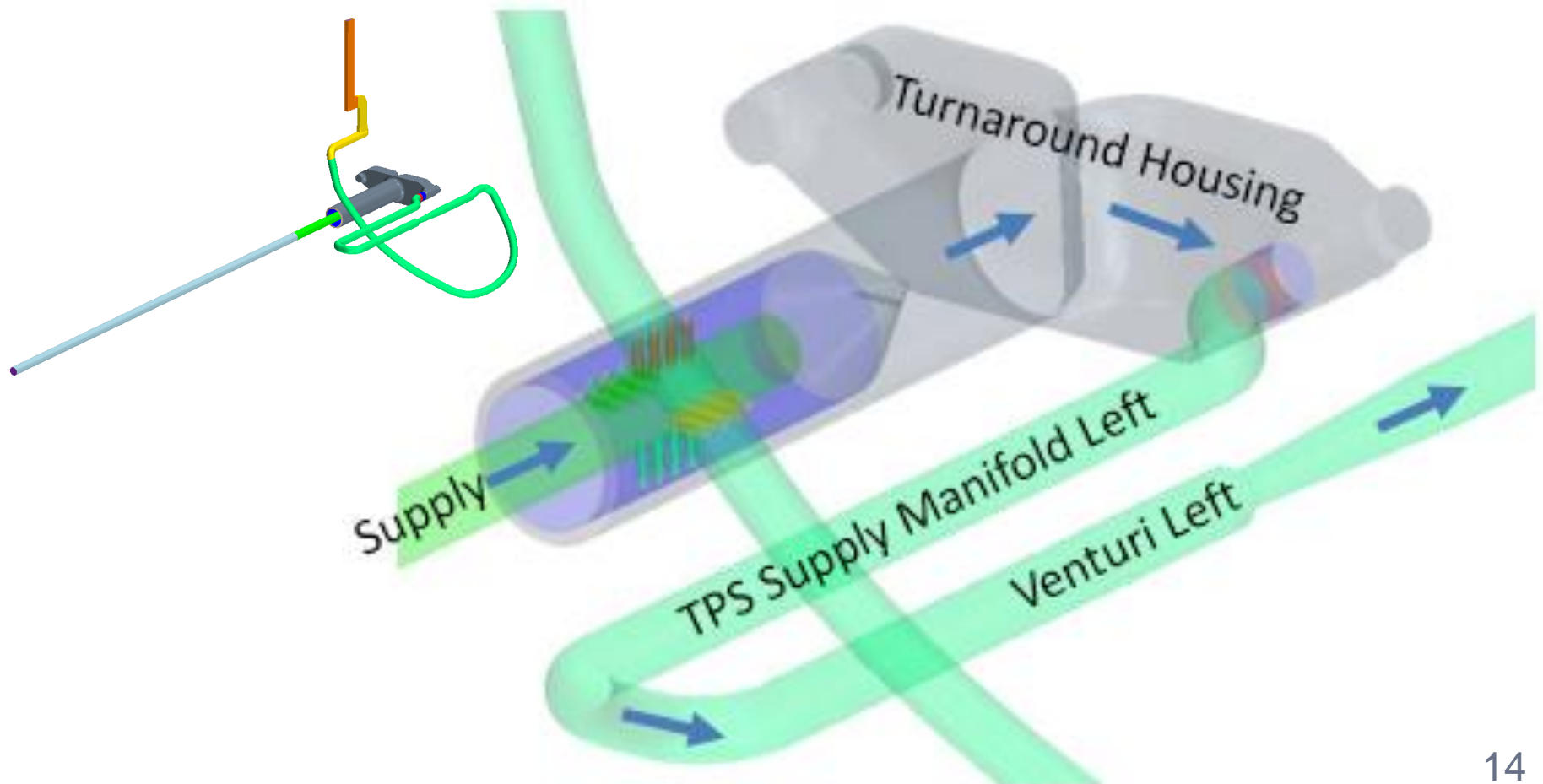
- CFD simulations with STAR-CCM+



- Single TPS unit calibration in 9x7 test section



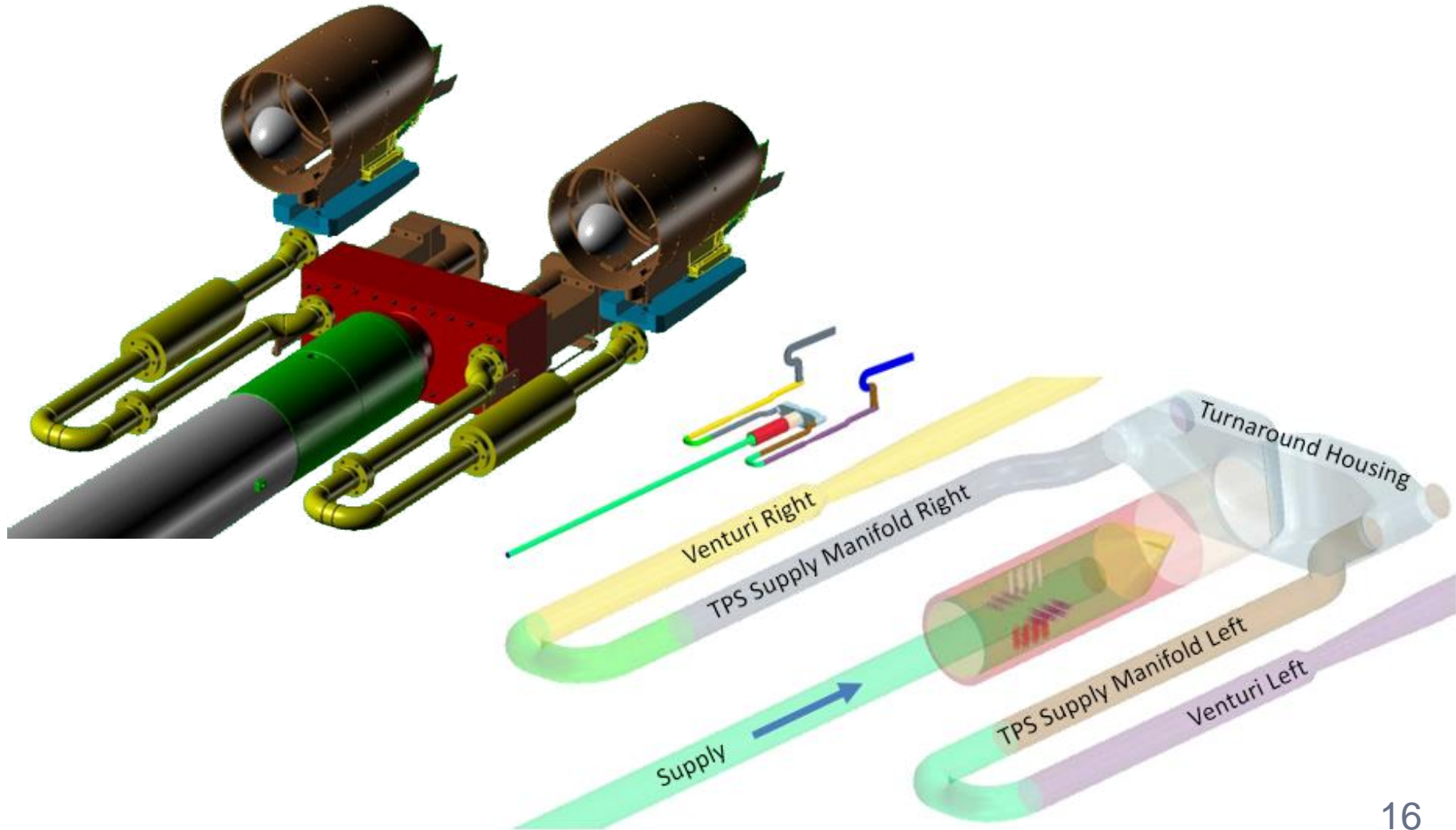
- Single TPS unit calibration in 9x7 test section



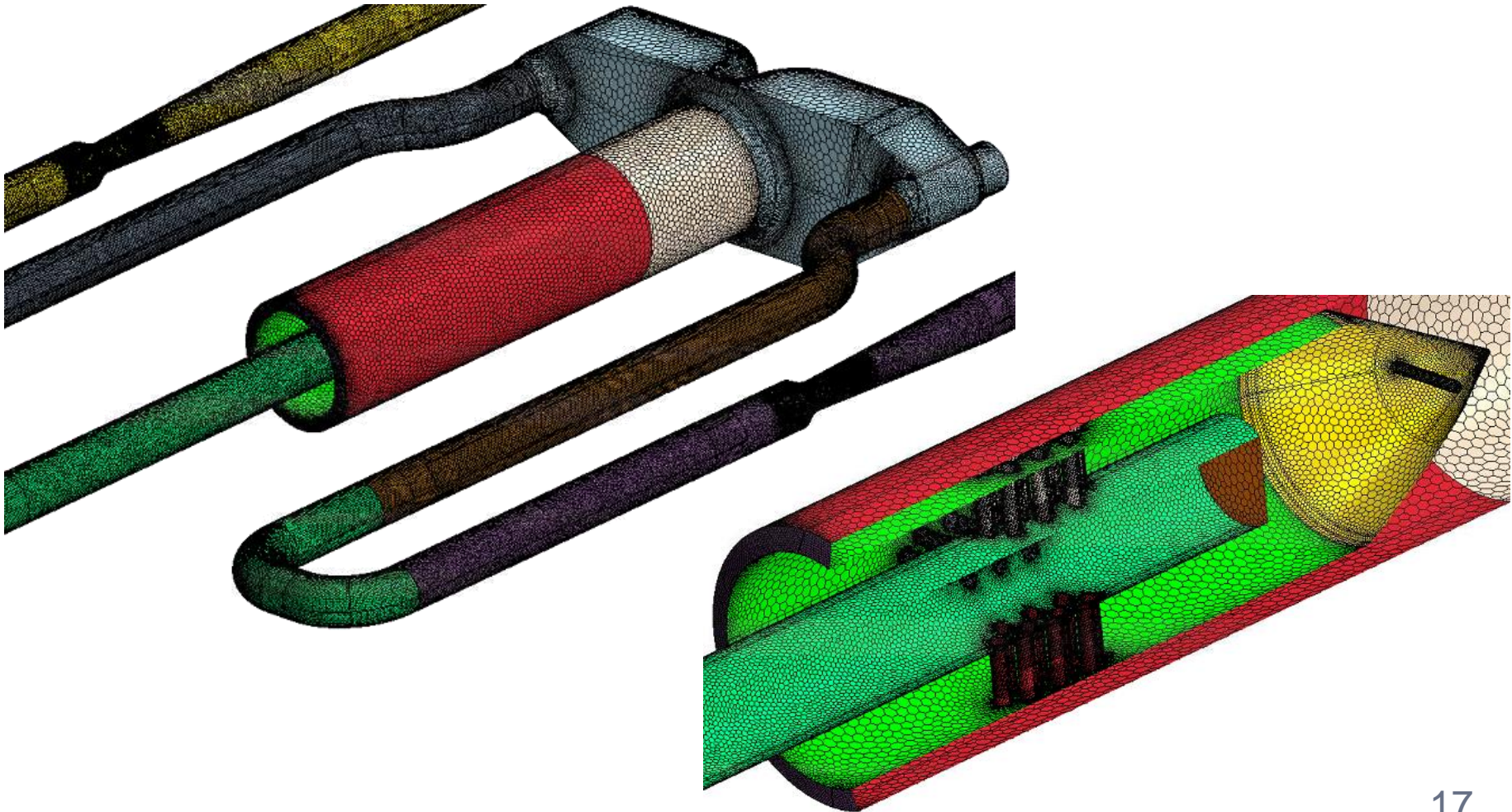
- Single TPS unit calibration in 9x7 test section

Exp. Supply Mass Flow (lbm/s)	Exp. Plenum Total Pressure (psi abs)	Exp. Plenum Total Temperature (F)	Exp. Venturi ΔP (psi)	Balance Axial Force (lbsf)	Balance Normal Force (lbsf)	CFD Supply Pressure (psi abs)	CFD Mass Flow (lbm/s)	CFD Plenum Total Pressure (psi abs)	CFD Plenum Total Temperature (F)	CFD Venturi ΔP (psi)	CFD Axial Tare (lbsf)	CFD Normal Tare (lbsf)
0.489	87.188	80.1	4.51	39.1	0.56	88.7	0.493	85.6	82	5.9	-0.4	50.8
1.509	279.14	89.7	14.6	134.2	2.48	283.7	1.55	274.4	90	18.3	-1.1	187.3
2.018	375.312	119.7	19.8	170.3	1.23	382.7	2.02	370.5	120	24.5	-1.7	256.9

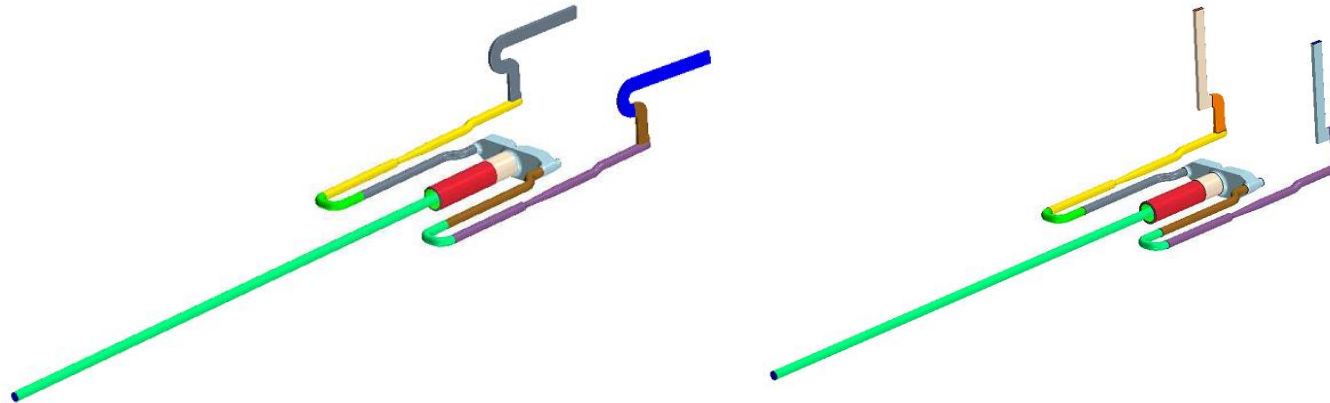
- Simulations for HWB with TPS in 40x80



- Simulations for HWB with TPS in 40x80



- Simulated two TPS exhaust directions



CFD Mass Flow (lbm/s)	CFD Plenum Total Pressure (psi_abs)	Balance Axial Force (lbf)	Balance Normal Force (lbf)	CFD: Hz Exit Axial Tare (lbf)	CFD: Hz Exit Normal Tare (lbf)	CFD: Vertical Exit Axial Tare (lbf)	CFD: Vertical Exit Normal Tare (lbf)	CFD: Hz Exit Axial Thrust (lbf)	CFD: Hz Exit Normal Thrust (lbf)
0.42	28.4	-	-	-19.5	-1.2	-1.5	18.2	19.2	1.24
0.9	65	35.4	-4.5	-77	-5.5	-5.3	69	77.1	4.95
3.4	245	221.0	60.6	-366	-21	-23.6	350	366	23.4

- Recommendations for future FTB testing
 - Buildup of manifold in concert with tares
 - Additional interior flowpath sensors
 - Accurate inlet and exhaust face measurements
 - Support and insight using CFD
- Acknowledgments
 - ARMD ERA Program
 - NAS Supercomputing Facility

